



Section 16: Impacts of the Microgravity Environment on Experiments

Richard DeLombard

Acceleration Measurement Discipline Scientist

NASA Glenn Research Center





Introduction

- Earth-bound experiments are affected by normal gravity and vibrational forces which exist in ground laboratories.
 - gravity, elevators, air conditioner, traffic, people
- Most microgravity experiments desire:
 - zero-gravity,
 - constant, uni-directional acceleration, and/or
 - constant conditions.
- Taking experiments to orbit removes effects of gravity but trades above disturbances for others
 - gravity gradient, aerodynamic drag, thrusters, other experiments, crew members, Shuttle subsystems





Improve the Environment

- Microgravity environment is close to "zero-g" conditions, but still precautions need to be taken.
 - operational inhibits
 - typical: attitude free-drift, crew exercise, & equipment operations
 - non-typical: Ku-band antenna & crew motion
 - Orbiter attitude and altitude requirements
 - Shuttle attitude determines the relative direction of the quasi-steady acceleration
 - slight attitude changes have effect on frequency of VRCS jet firing
 - amount of deadband in attitude control has effect on frequency of VRCS jet firing
 - Shuttle altitude has effect on the magnitude of the drag component and fluctuations thereof, especially if orbit is not circular





Microgravity Science Experiment Successes

- Vast majority of microgravity science experiments have been successful
 - success has been due to a number of factors
 - prior ground based research
 - PI involvement and extensive review process during experiment design and development
 - mission planning and operational simulations
 - crew interest and training
 - support for PI teams during mission
 - microgravity environment which can be supplied by the Shuttles, Mir, parabolic flight aircraft, sounding rockets, and free flyers
- Some problems have occurred with experiments
 - · difficulties with a few factors
 - hardware failures on-orbit
 - modified mission parameters
 - unexpected operational scenarios
 - microgravity environment violated requirements





Advanced Automated Directional Solidification Furnace on USMP-2 / Lehoczky

- High temperature furnace
 - five zones, Bridgman-Stockbarger-type furnace
 - reduced stratification and fluid flow desired in microgravity
 - three Orbiter attitudes created different quasi-steady acceleration levels at furnace location
- Two attitudes presented unstable conditions
 - acceleration aligned with sample cold-to-hot
 - convective situation liable to lead to fluid flow
- Perpendicular / transverse acceleration ratio
 - · less than one for all three attitudes
 - also conducive for fluid flow





AADSF on USMP-3 / Fripp

- Objective of experiment:
 - The primary objective of this flight experiment was to examine the effect of the direction of the microgravity vector on the convective mixing of the liquid during directional solidification.
- Method:
 - three sample cells
 - three Shuttle attitudes (one for each cell)
 - thermally stable but solutally unstable in a one dimensional analysis
 - solutally stable but thermally unstable in a one dimensional analysis
 - horizontal growth
- Unplanned attitude change
 - cell #1 processing interrupted for free drift
 - acceleration vector alignment angle went from 2° to 45°
 - result appears to be solidification in accordance with a fully mixed condition as opposed to the desired diffusion condition





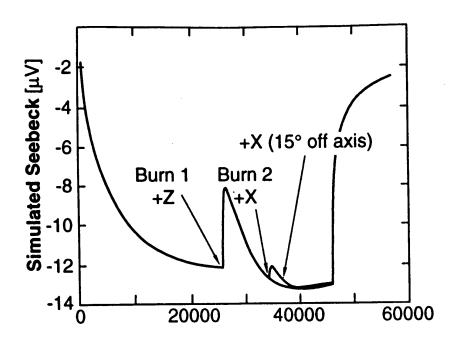
MEPHISTO on USMP-3 / Favier

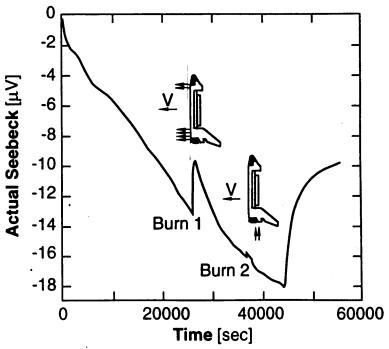
- High temperature furnace
 - metal alloy samples
 - in-situ measurement of Seebeck voltage indicating average melt composition at interface
- One objective was to quantitatively characterize microgravity effects on an actual crystal growth experiment
- Thruster firing effects
 - high acceleration levels cause convective mixing
 - interfacial composition disturbed
- Summary results
 - PRCS thruster: approximately one hour to recover from short thruster firing of 10 to 25 seconds durations
 - OMS thruster: approximately sixteen minutes to recover from short (35 seconds) duration OMS firing





Response to PRCS & OMS Thruster Burns









Indium Float-zone Furnace on STS-32 / Dunbar & Thomas

- Fluid Experiment Apparatus
 - float-zone materials processing furnace in middeck
 - Indium sample
 - Eigen-frequencies in 0.1 to 10 Hz range
 - crew treadmill exercise and thruster firing disturbances
- Acceleration measurements made with Honeywell In-Space Accelerometer (HISA)
 - mounted on front of Fluid Experiment Apparatus





Movement of Particles on IML-1 / Trolinger

- Movement of particles with different densities
 - secondary science experiment within the Fluids Experiment System facility to grow triglycine sulfate crystals
 - tracer particles of three sizes (small, medium, large)
 - tracer particle motion was studied relative to fluid convection and the microgravity environment
 - correlations established between particle motion and acceleration environment
 - the growth solution is still far from the static system originally envisioned by microgravity scientists





GaAs crystals during USML-1 / Matthiesen

- Experiment conducted in the Crystal Growth Furnace
 - examined the radial dopant distribution with the quasi-steady microgravity vector
 - microgravity vector acted perpendicular to the growth direction (i.e. transverse acceleration)

Results

- microgravity environment was quantitatively linked to the quality of the crystal
- complete-mixing profile of the crystal was caused by the crucible-melt interface shape and not by the microgravity environment





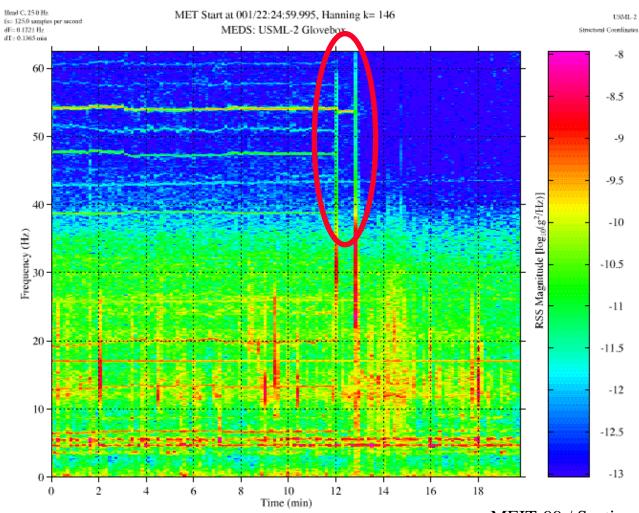
Surface Tension Driven Convection Experiment on USML-2 / Ostrach

- STDCE sample container
 - free liquid surface of silicone oil
 - study of surface tension driven convection with different fill levels
- Disturbances noted during operations
 - ripples on free liquid surface
 - correlation established between glovebox fan operation, experiment video downlink, and downlink acceleration data
 - disturbance transmitted along and across aisle of Spacelab module





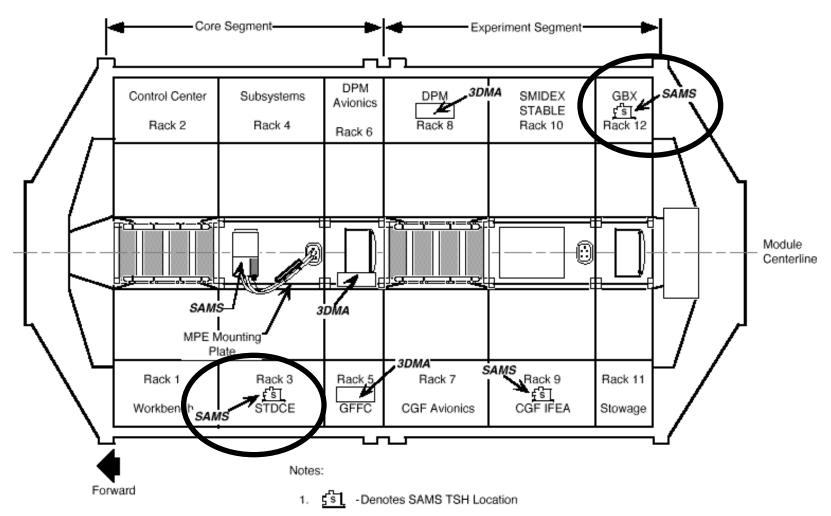
USML-2 Glovebox Fan De-activation







Location of STDCE and Glovebox on USML-2







Confined Helium Experiment on USMP-4 / Lipa

- Liquid helium sample
 - temperature about 2 K
 - low mass
- Pre-mission concern about structural resonances
 - narrow frequency bands around 55, 75, & 119 Hz
 - 3rd harmonic of Ku-band antenna dither (~51 Hz) was a particular concern
 - Unexpected disturbance was observed around 56 Hz during initial part of mission
 - evaluated SAMS data in real-time and off-line to determine source





USMP-4 Inter-Experiment Disturbance

- Early in the mission, an unexpected signature appeared around 56 Hz
- PIMS notified the CHeX team of the observation, as well as the other USMP-4 science team via flight notes
- Near real-time plots were provided to the CHeX team in an effort to characterize this disturbance source
 - Based on the stability observed in these plots, the CHeX team was able to adequately compensate the CHeX data for this disturbance
 - Analysis of the activation times for the other USMP-4 payloads indicated the IDGE experiment introduced the 56 Hz disturbance to the environment upon their activation





USMP-4 Disturbance Source Identification

- Source of the 56 Hz disturbance was determined to be IDGE cooling fans
- Additional disturbance sources at 37 Hz and 74 Hz were introduced by the IDGE experiment hardware
- Although IDGE had flown on previous USMP missions, the disturbances were unknown because of lower cutoff frequencies for SAMS sensors flown on those previous USMP missions
- Post-mission ground test with IDGE flight experiment hardware and SAMS flight hardware confirmed the flight data conclusions





USMP-4 Operations Dodged a Bullet

- One of the IDGE cooling fans operated in the 37 Hz region
 - The operating speed of this fan was variable
 - Thus the vibration disturbance was a varying frequency
- LPE compensation
 - A variable frequency disturbance near a structural mode of LPE would have been very difficult or impossible to compensate
 - Likely would have resulted in curtailed operation of LPE, IDGE, or both during USMP-4





Protein Crystal Growth / various Pls

- Nominal operations require quiet vibratory acceleration environment (<10⁻⁴ g) for first 24 - 48 hours for crystal nucleation
- On missions with significant disturbances, showers of small crystals have been observed
- Correlations between growth rate and crew exercise periods have been observed



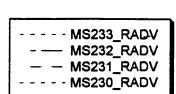


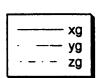
Structure of Flameballs at Low Lewis Numbers on STS-94 / Ronney

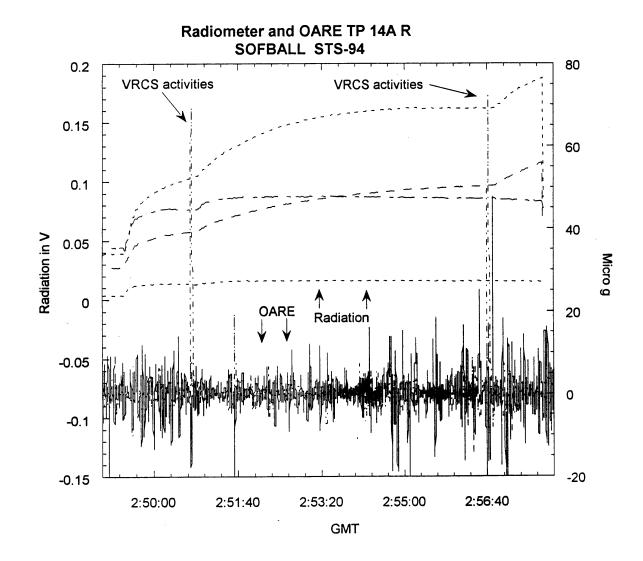
- Thruster (VRCS jets) disturbances caused significant, longlasting perturbations to the radiation readings
 - effect appears to last several minutes
- PI did not request free-drift periods pre-mission
 - need became apparent during early experiment operations
 - another PI team had a pre-mission "no-free drift" requirement
 - SOFBALL monitored the location of the Orbiter within the attitude deadband and initiated operations accordingly
 - Achieved maximum 5-minutes in deadband drift at beginning of their 14 minute test sequence
- PI is investigating the cause
 - flame ball surrounded by much larger region (a few cm in diameter) of hot gas - very sensitive to any acceleration
 - radiometers respond sharply when acceleration deforms the hot gas and shifts closer to or further away from the radiometer















Structure of Flameballs at Low Lewis Numbers (SOFBALL) / Ronney

- Planned re-flight of SOFBALL on STS-107
- PI included requirements for low-frequency acceleration measurements
- PI has re-affirmed these requirements during PAYLOAD manifesting negotiations
 - Initial difficulty in manifesting OARE accelerometer
 - PI's requirements ensured that OARE would be manifested





Isothermal Dendritic Growth Experiment on USMP-2 & -3 / Glicksman

- For all practical purposes, with respect to dendritic growth for the chamber size and temperature range selected, the microgravity environment aboard the Shuttle is at "zero-g," i.e., convective effects are not seen on-orbit.
 - convection exists at 1 g
 - convection not apparent at 1 μg
 - convection not apparent at 40 μ g during Tether Satellite System deployment on STS-75
 - no disturbance for 40 -> 0.5 μg transition when TSS broke
 - threshold somewhere between 1 μ g and 1 g but apparently above the Shuttle environment





Summary

- The microgravity environment is not "zero-g" or even "zero-acceleration". It is dynamic.
- The microgravity environment may influence the results of a science experiment.
- Analyses and/or tests should be performed before flight to investigate the sensitivity of an experiment to the microgravity environment.
- Environments of past missions should be considered in planning future experiments.
- Experiment teams should be concerned about what disturbances they may be causing to the microgravity environment with (for example) moving parts or required crew actions.





References

- Dunbar, B. J., R. L. Giesecke, and D. A. Thomas: "The Microgravity Environment of the Space Shuttle Columbia Payload Bay During STS-32." NASA TP-3141, 1991.
- Schoess, J. N., B. J. Dunbar, D. A. Thomas: "Microgravity Environment Measurements On Board Space Shuttle Columbia on STS-32." SENSORS, Vol. 7, No. 11, October 1990, pp. 15-19.
- Rogers, M. J. B., J. I. D. Alexander, and Schoess, J.: "Detailed Analysis of Honeywell In-space Accelerometer Data - STS-32." Microgravity Science and Technology, vol. VI, issue 1, pp. 28 - 33, 1993
- Ramachandran, N., Frazier, D. O., Lehoczky, S. L., and Baugher, C. R.: "Joint Launch
 + One Year Science Review of USML-1 and USMP-1 with the Microgravity
 Measurement Group." NASA CP-3272, 1994
- Curreri, P. A. and McCauley, D. E.: "Second United States Microgravity Payload: One Year Report." NASA TM-4737, 1996
- Fripp, A. L., Debnam, W. J., Rosch, W. R., and Narayanan, R.: "The Effect of Microgravity Direction on the Growth of PbSnTe." Launch + One Year Report (USMP-3), 1997





References (cont'd)

- Alexander, J. I. D., Garandet, J. P., Favier, J. J., and Lizee, A.: "Quantitative Experimental Characterization of g-jitter Effects on Directional Solidification." AIAA97-0675, 1997
- Rogers, M. J. B. and DeLombard, R.: "Summary Report of Mission Acceleration Measurements for STS-73." NASA TM-107269, 1996
- Snell, E. H.; Boggon, T. J.; Helliwell, J. R.; Moskowitz, M. E.; and Nadarajah, A.: "CCD video observation of microgravity crystallization of lysozyme and correlation with accelerometer data." Submitted to Acta Cryst. D., 1996.